



# **WFIRST Low Order Wavefront Sensing and Control Testbed Performance Under Flight Like Photon Flux**

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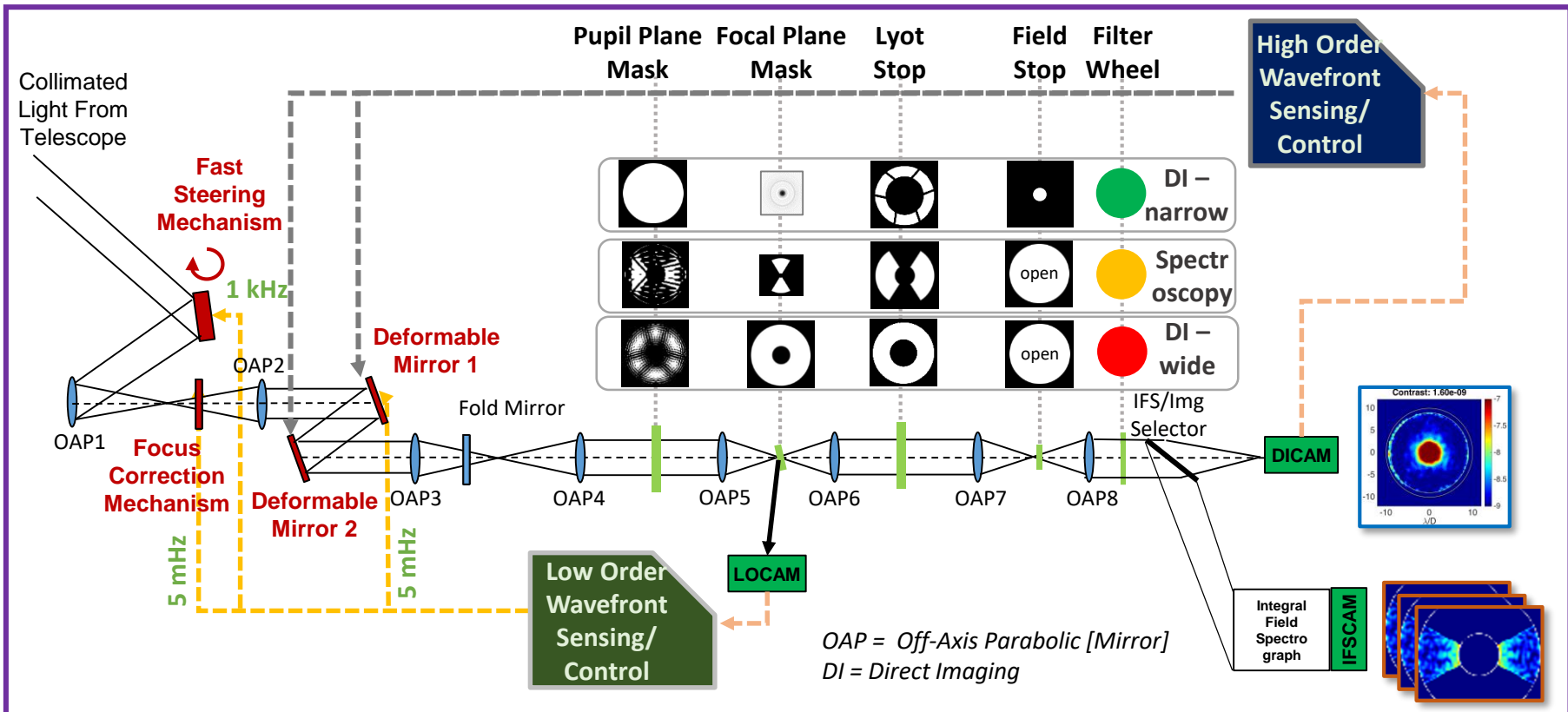
The decision to implement the WFIRST mission will not be finalized until NASA's completion of the National Environmental Policy Act (NEPA) process. This document is being made available for information purposes only.

- Overview of Low order wavefront sensing and control (LOWFS/C) for WFIRST Coronagraph Instrument (CGI)
- Testing LOWFS/C at flight like photon flux
- LOWFS/C line-of-sight sensing and FSM control performance
- LOWFS/C focus sensing and DM correction performance
- Conclusion and future work

Companion paper:

10698-95: Hybrid Lyot coronagraph for WFIRST: high contrast testbed demonstration in flight-like environment, Byoung-Joon Seo, et al

# WFIRST CGI LOWFS/C Overview

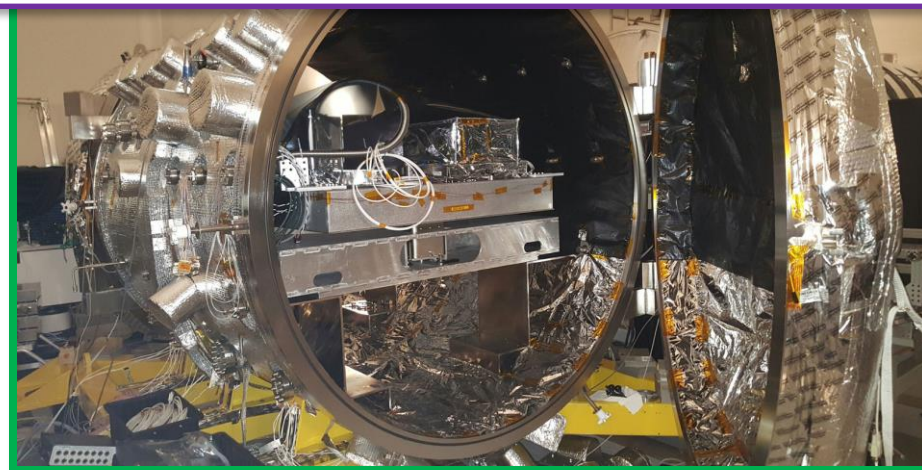
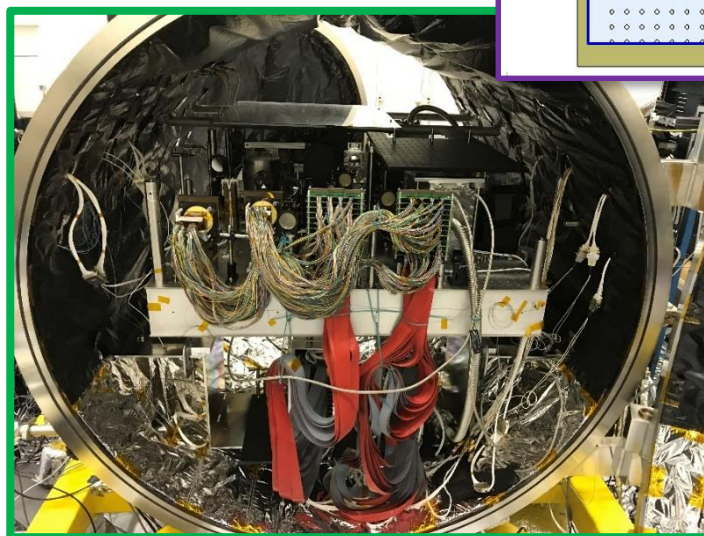
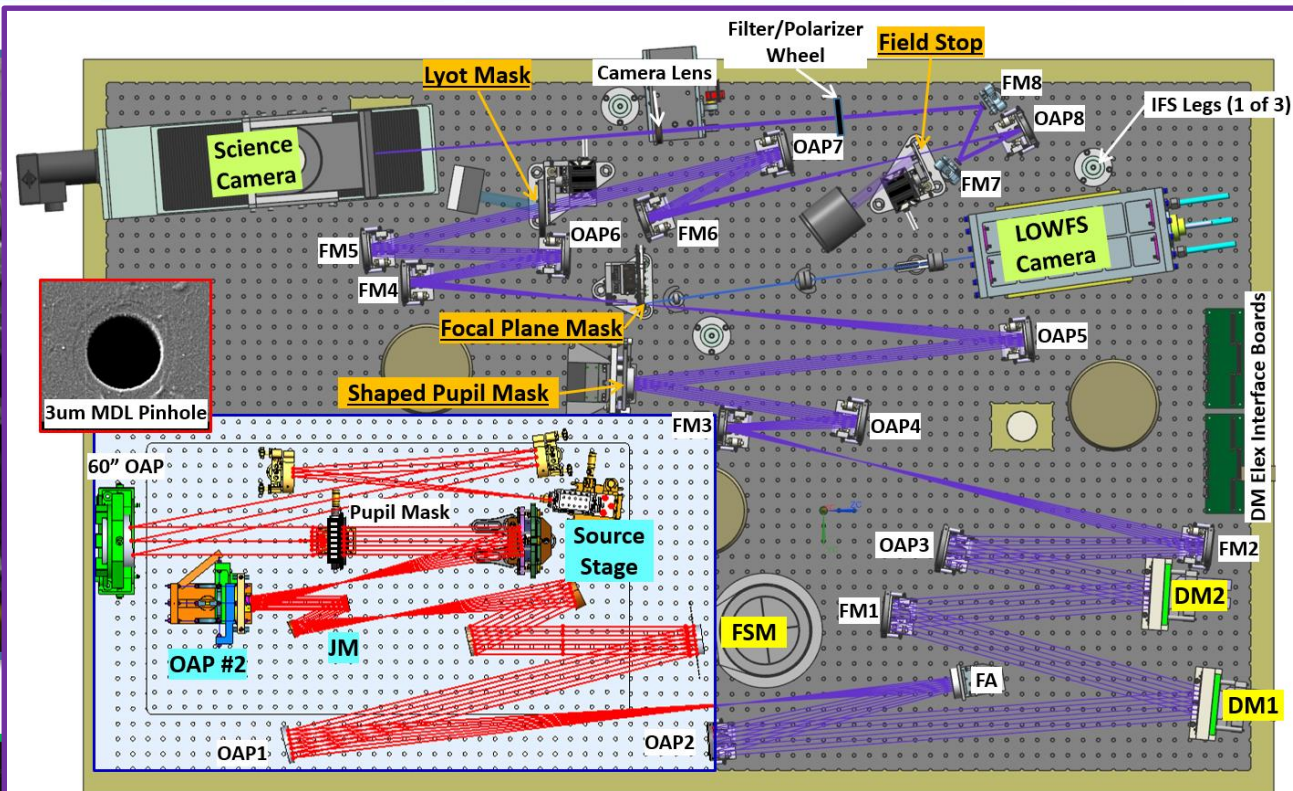
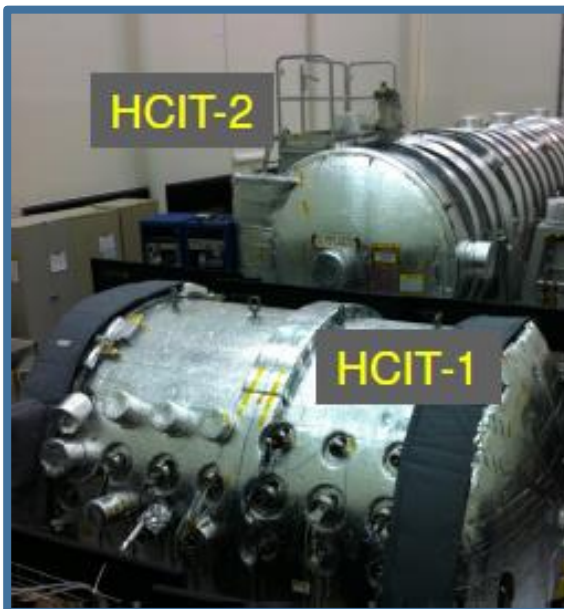


- LOWFS/C subsystem measures and controls line-of-sight (LoS) drift and jitter as well as the thermally induced low order wavefront drift. LOWFS sensor is Zernike wavefront sensor (ZWFS).
  - LoS: drift ( $< 2$  Hz):  $\sim 14$  mas, tonal jitter:  $\leq 14$  mas
  - WFE: drift ( $\sim 10^{-3}$  Hz):  $\sim 0.5$  nm (RMS), dominant by focus, astigmatisms and comas from the telescope optics rigid body motions
- Uses rejected starlight from occulter which reduces non-common path error
- LOWFS is a differential image wavefront sensor referenced to star light suppression wavefront control (HOWFS/C): it maintains wavefront established for high contrast





# Occulting Mask Coronagraph (OMC) Dynamic Testbed



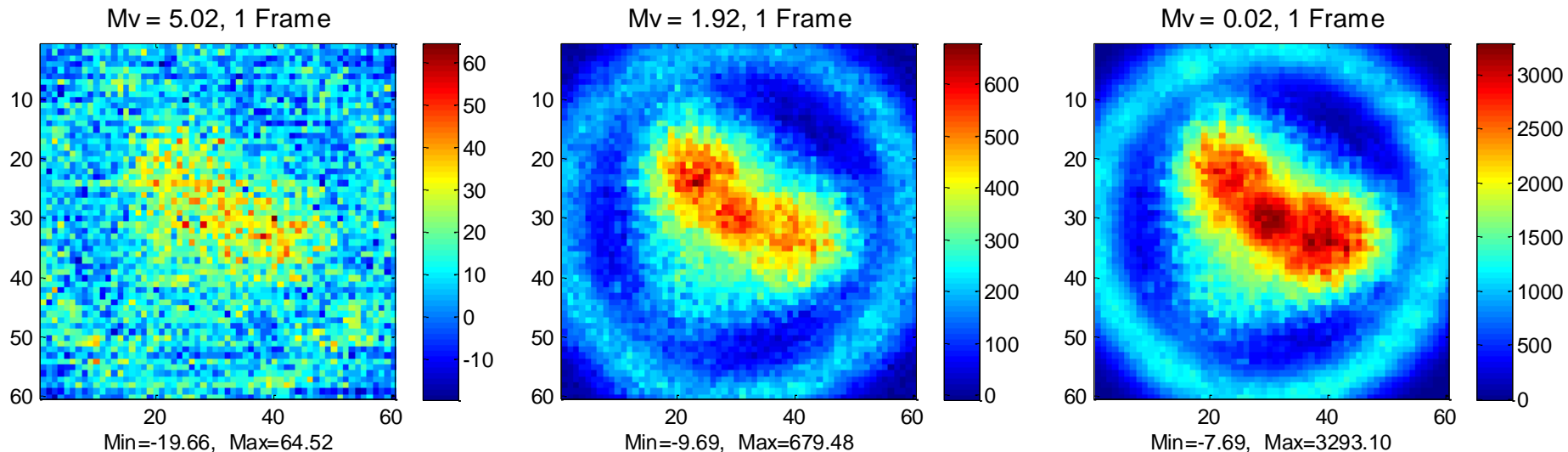
# LOWFS/C Test with Flight Like Wavefront Disturbances

- In our previous dynamics test (2017) we have demonstrated that LOWFS/C can maintain CGI contrast stability to better than  $10^{-8}$  in presence of WFIRST like LoS and low order WFE disturbances in both SPC and HLC modes
  - Three wavefront aberration modes demonstrated (tip-tilt and focus) are the dominant disturbances for WFIRST Coronagraph
  - LOWFS/C LoS control using the FSM and low order wavefront correction using a DM were demonstrated.
  - However, these tests were done on testbed using a bright source which has a brightness equivalent to a  $M_v = -3.5$  star.
- What will the LOWFS/C perform be under a realistic photon flux expected during the WFIRST CGI on-sky observation? The baseline WFIRST requirement for LOWFS/C on star brightness:
  - Maintain the wavefront stability during CGI initial star light suppression wavefront control (EFC) on stars with brightness  $M_v \leq 2.0$
  - Maintain the wavefront stability during CGI science target observations (no EFC) on stars with brightness  $M_v \leq 5.0$
- LOWFS/C needs to perform under flight like photon flux



# LOWFS/C Test with Flight Like Photon Flux and Disturbances

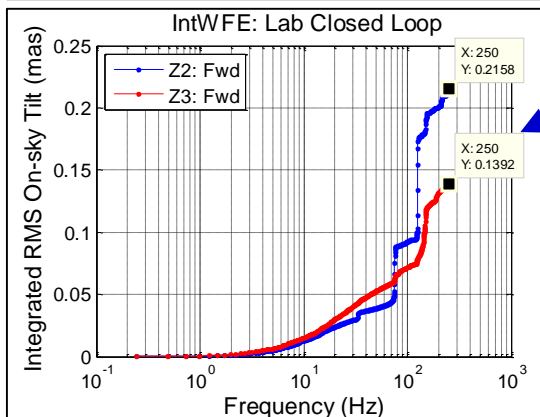
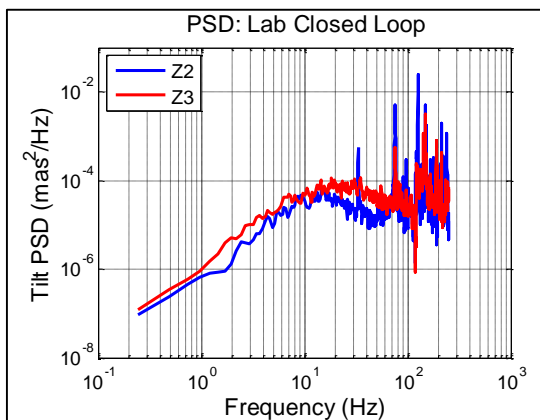
- Photon flux on the testbed LOWFS camera is measured and calibrated to equivalent total photoelectrons per frame for WFIRST LOWFS camera.
- Combination of ND filter and source power are used to reduce the photon flux on the LOWFS camera while fix source spectrum to the design bandwidth
  - LOWFS camera exposure time is fixed for the high speed read out
- Examples of a single frame image from testbed's LOWFS camera at equivalent stellar magnitude of  $M_v = 5$  (left),  $M_v = 2$  (middle), and  $M_v = 0$  (right)
- **Fainter star image has lower image count (DN) and the photon noise is more pronounced**



# Compare Testbed Low Flux Sensing Error Against Sensor Model Prediction

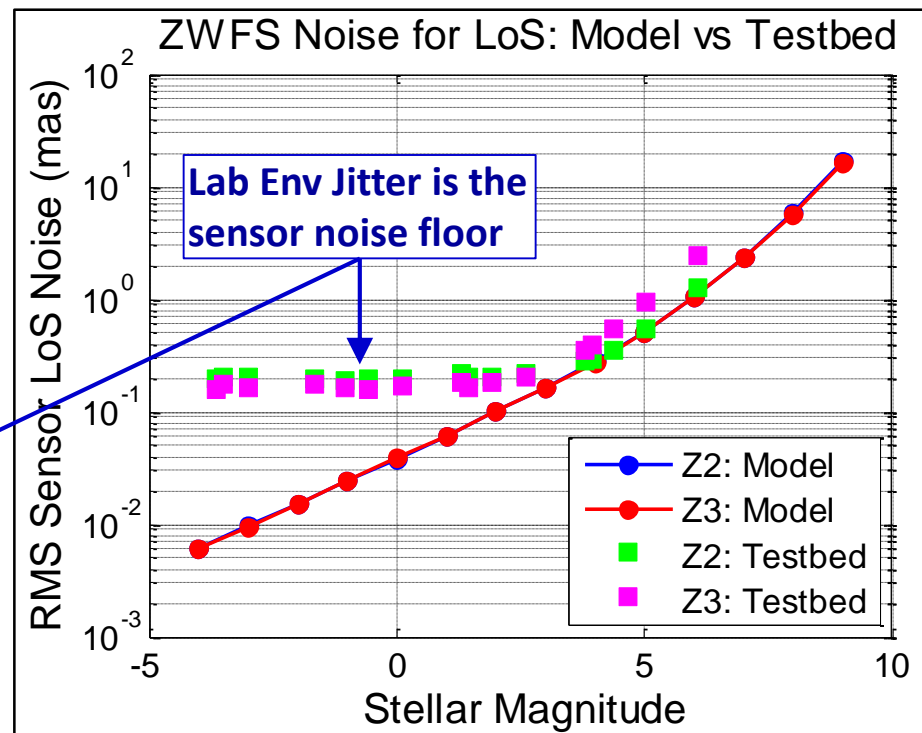
## • Testbed setup

- Under lab environment with FSM loop closed (PSD and integrated WFE plots on the left).
- Testbed measurement includes the lab environment jitters:  $\sim 0.2$  mas
- Testbed CMOS camera read out noise is different in row and column directions, causing the difference of X and Y sensing noise.



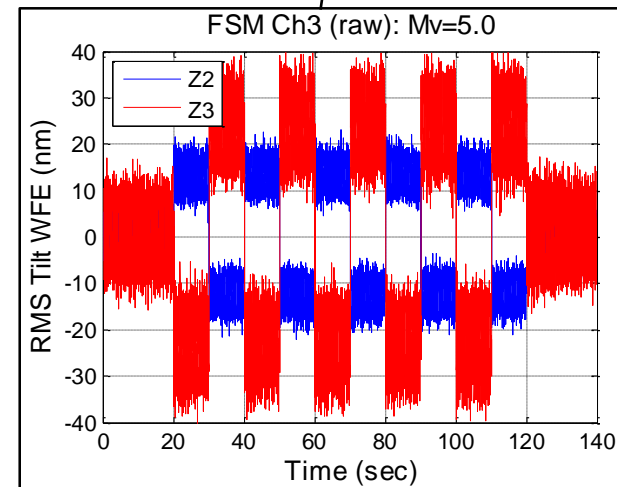
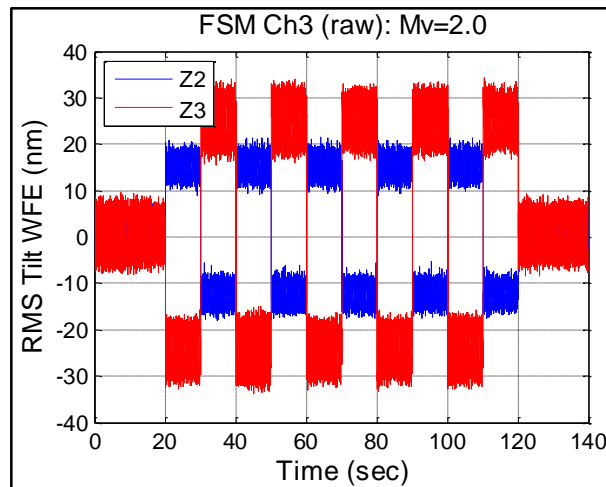
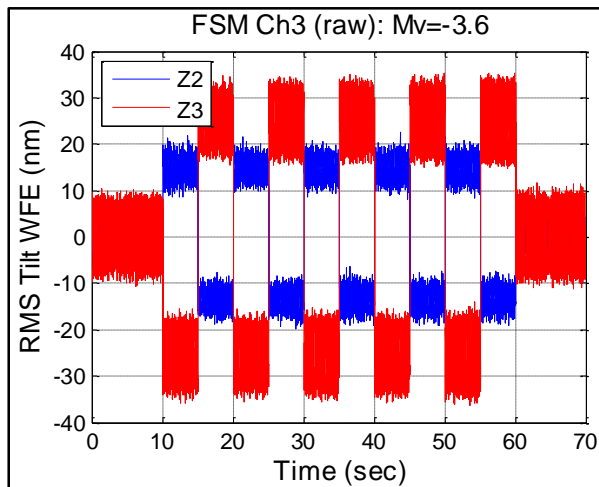
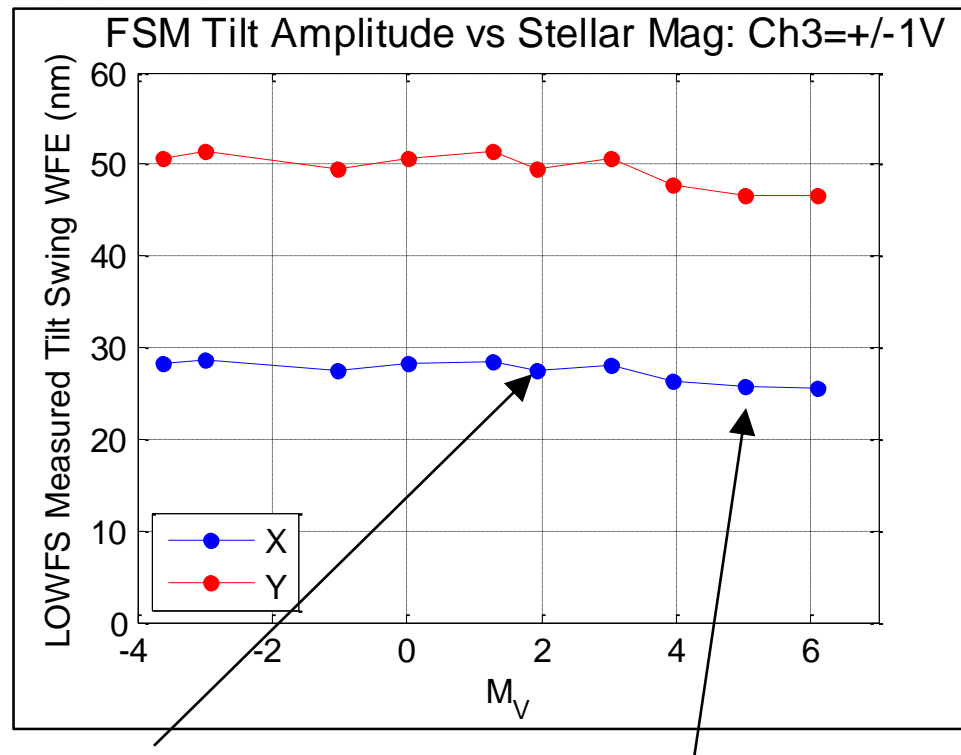
## • LOWFS sensor model:

- Use testbed LOWFS camera image sampling
- Including photon noise, read out noise, dark current
- Model curve show that sensor error is dominated by photon noise for bright stars ( $M_v < 3$ )
- Testbed data matches model prediction at low flux region, where sensor noise is dominated
- At high flux region the measured error is limited by the lab environment jitter



# LOWFS Sensing Accuracy vs. Stellar Magnitude: Line-of-Sight (Z2 & Z3)

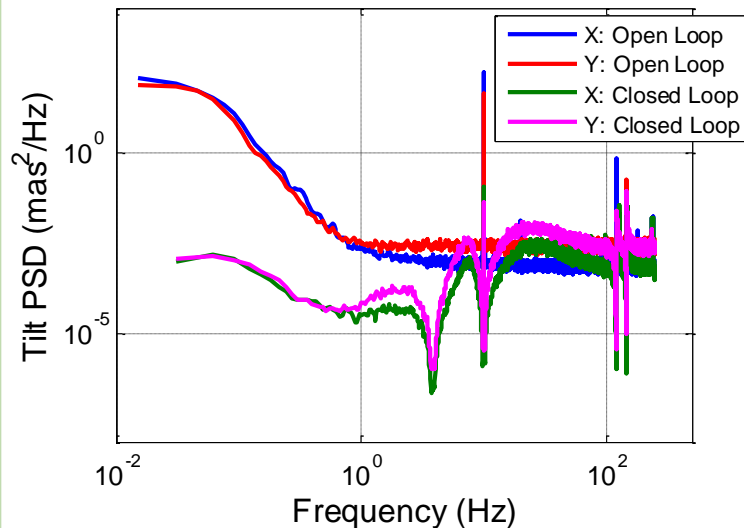
- Fixed voltage square wave command is applied to FSM actuators, creating a fixed tilt wave form (Ch3 data are plotted)
- LOWFS sensor measured the chopping amplitude are compared.
- LOWFS tilt measurement remains fair constant through many orders of magnitude ( $10^4\times$ ) of source brightness.**





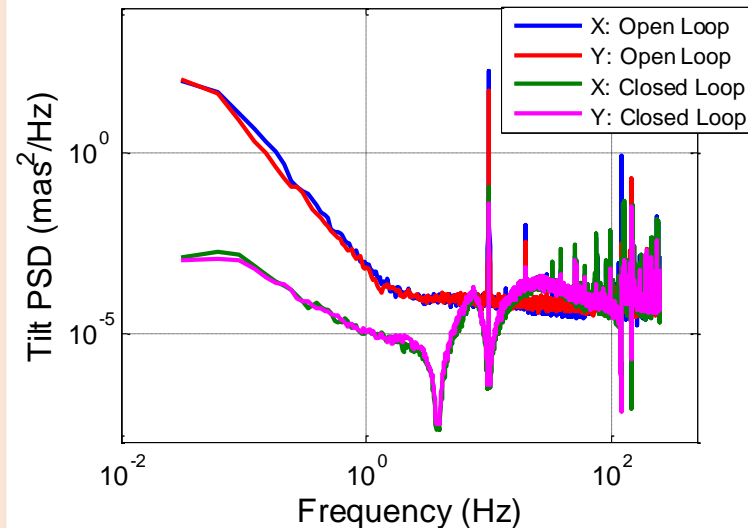
## Performance on Mv = 5 Star

Open&Closed Loop on JM Disturbances: Mv=5

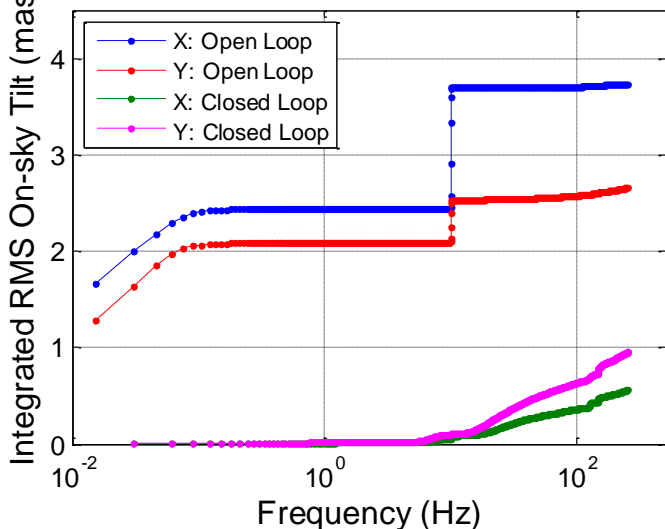


## Performance on Mv = 2.5 Star

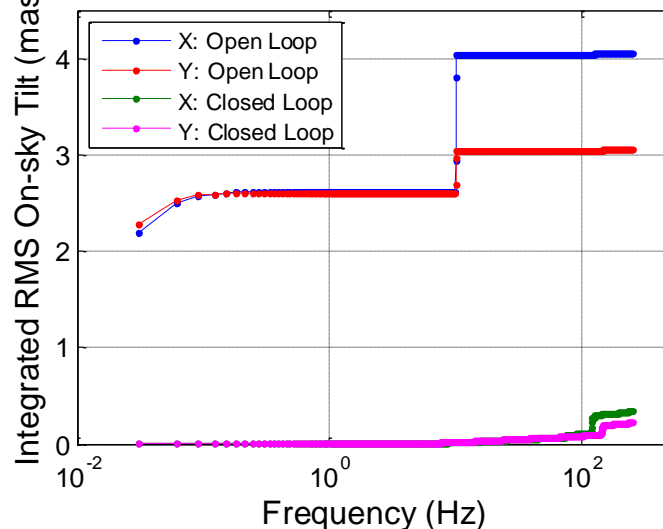
Open&Closed Loop on JM Disturbances: Mv=2.5

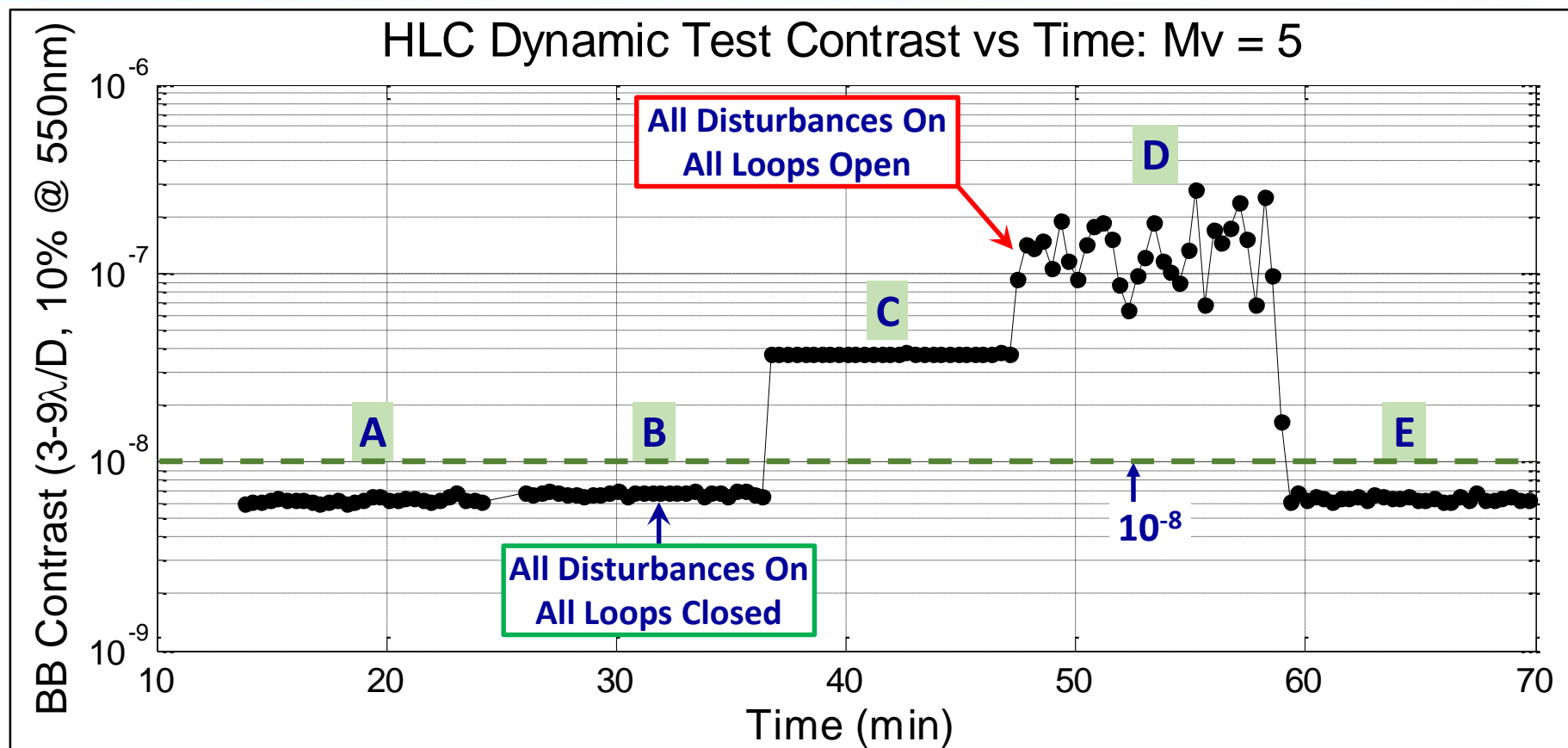


Open&Closed Loop on JM Disturbances: Mv=5



Open&Closed Loop on JM Disturbances: Mv=2.5

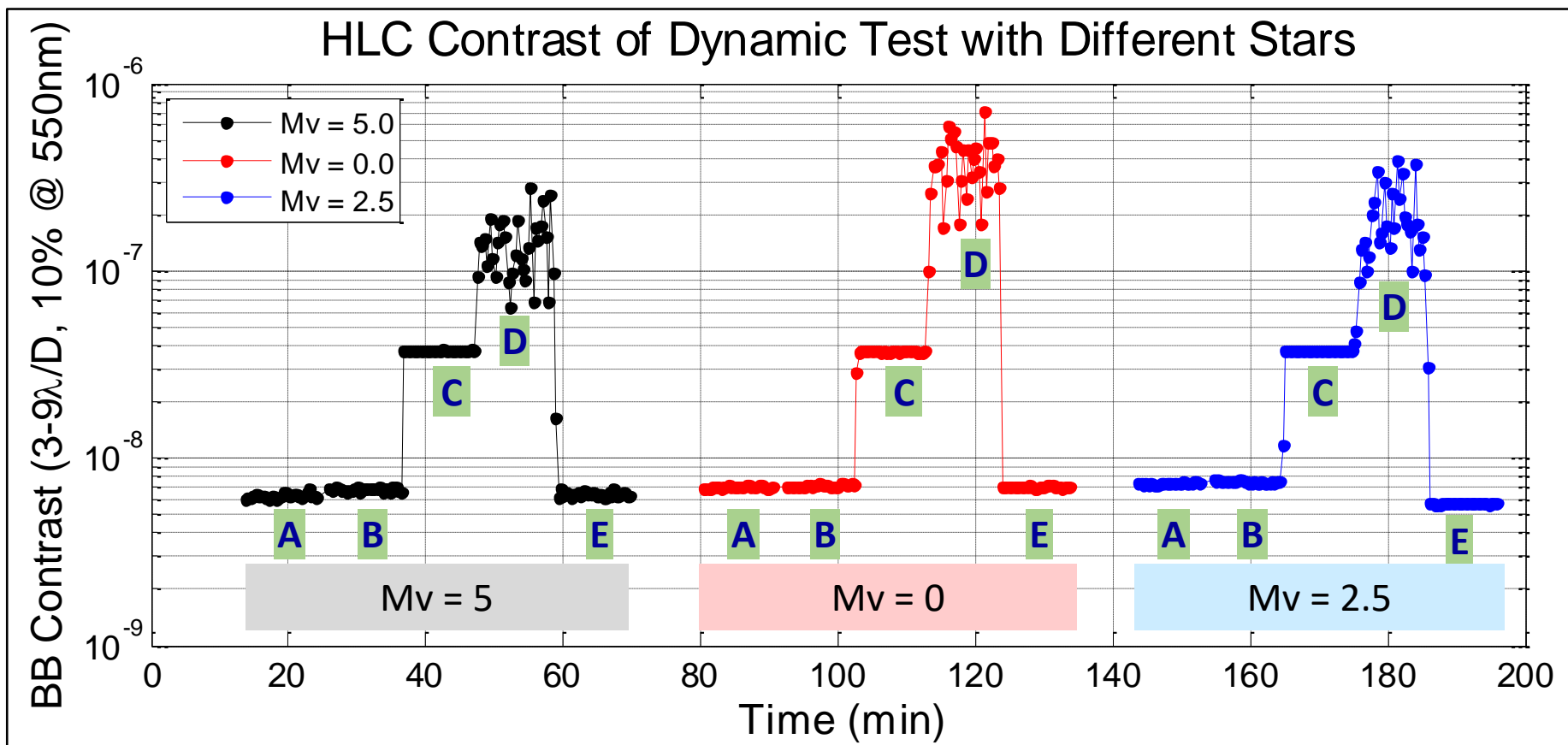




Sequences of dynamic test (each ~10 min):

- A. FB on & FF on with lab environment
- B. FB on & FF on with JM induced dynamics (ACS + RWA jitter at 600rpm)
- C. FB on & **FF off** with JM induced dynamics (ACS + RWA jitter at 600rpm)
- D. **FB off** & **FF off** with JM induced dynamics (ACS + RWA jitter at 600rpm)
- E. FB on & FF on with lab environment

# LOWFS/C LoS Dynamic Test with Different Stellar Magnitudes

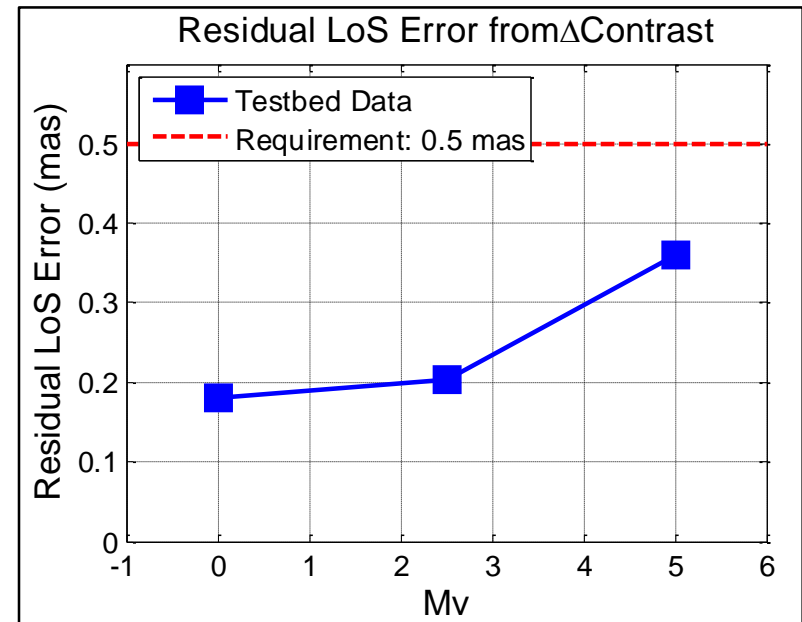
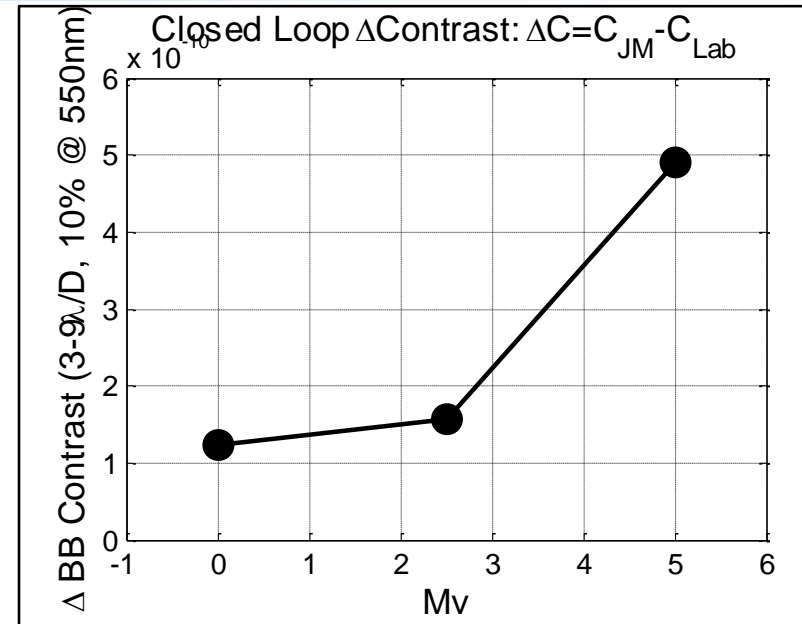


Sequences with each stellar magnitude ( $M_v = 5.0$ ,  $M_v = 0.0$ ,  $M_v = 2.5$ ):

- A. FB on & FF on with lab environment
- B. FB on & FF on with JM induced dynamics (ACS + RWA jitter at 600rpm)
- C. FB on & **FF off** with JM induced dynamics (ACS + RWA jitter at 600rpm)
- D. **FB off** & **FF off** with JM induced dynamics (ACS + RWA jitter at 600rpm)
- E. FB on & FF on with lab environment

# Post LoS Correction Residual Error Calculated from $\Delta$ Contrast

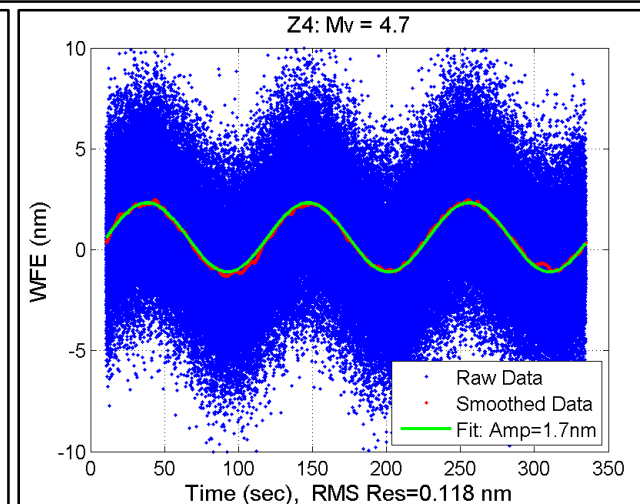
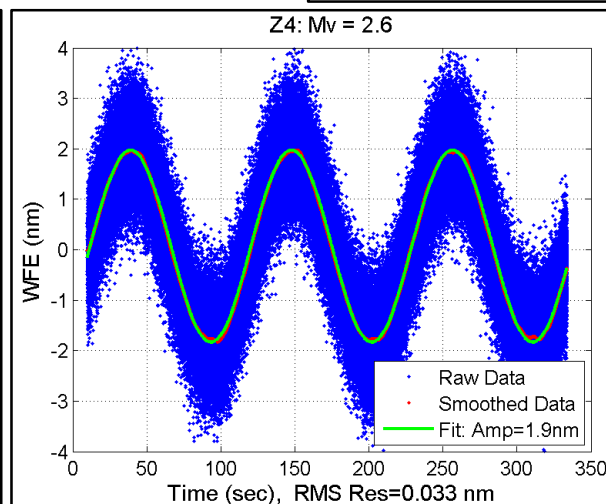
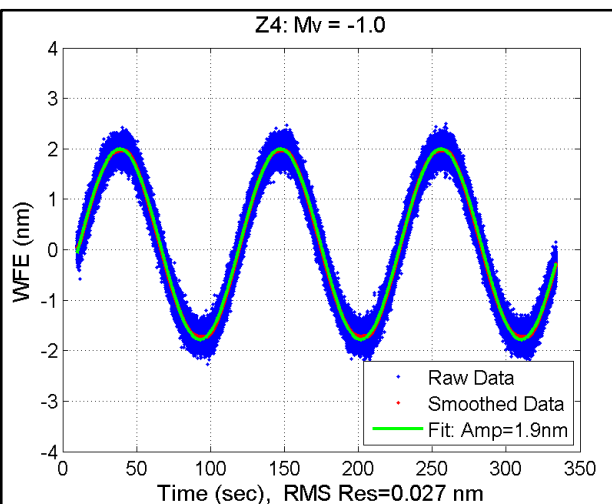
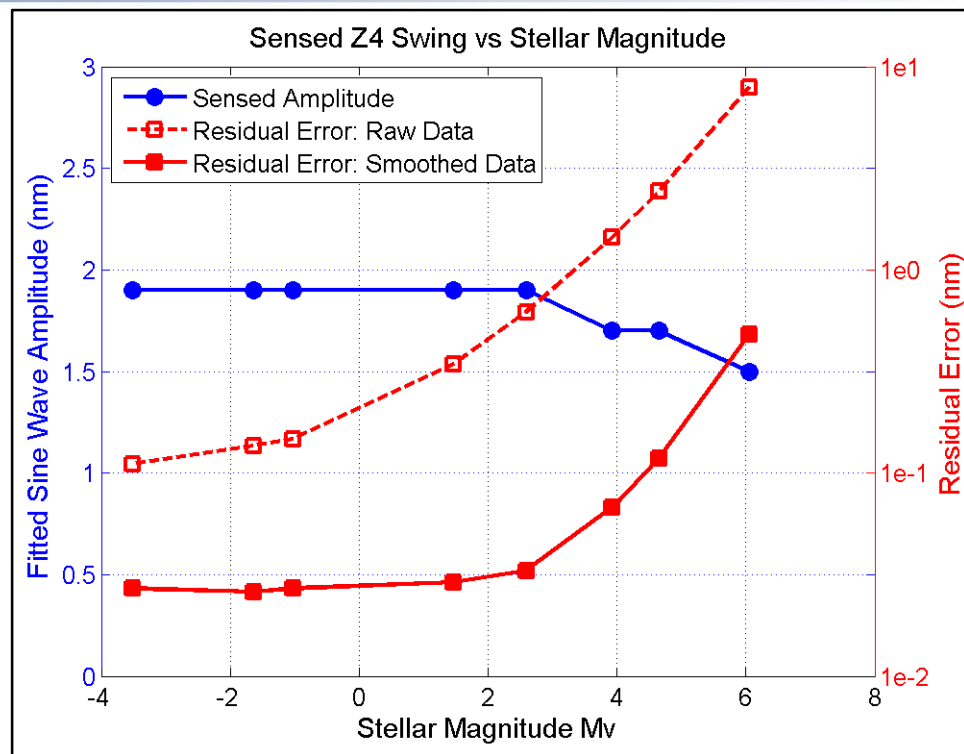
- Delta contrast ( $\Delta C$ ) between the closed loop at lab environment (quiet) and JM disturbances ( $\Delta C = C_{JM} - C_{lab}$ ) shows the effectiveness of LOWFS/C LoS loop suppression of JM induced WFIRST like LoS disturbances.
  - In previous plots Mean of B – Mean of A
- Fainter star has more contribution of contrast degradation due to the larger LOWFS sensing error (upper right plot)
- Using HLC jitter sensitivity we can calculate the post correction residual LoS error
- Convert  $\Delta C$  to LoS residual error:
  - Use contrast sensitivity from testbed:  
 $1.89e-9 \Delta C / \text{mas}^2 (3-9 \lambda/D)$  per axis
- The results ( $\leq 0.36$ ) met the requirement of 0.5 mas residual even at fainter star ( $Mv=5$ )**



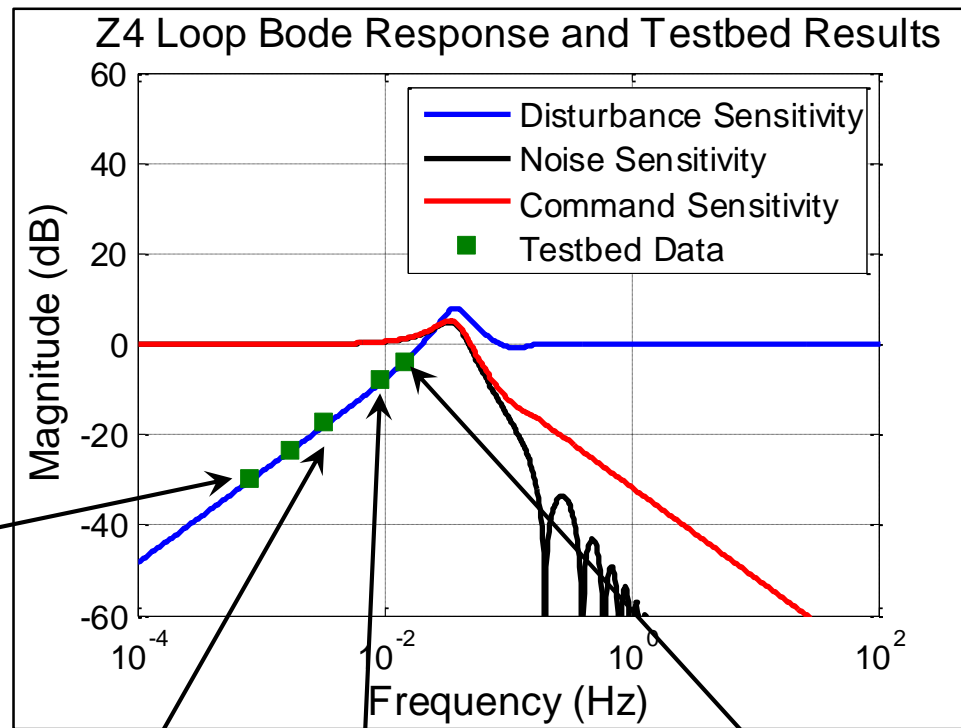
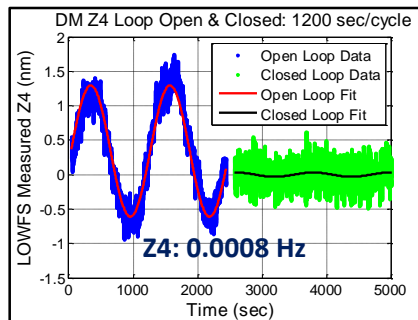


# LOWFS Sensing Accuracy vs. Stellar Magnitude: Focus (Z4)

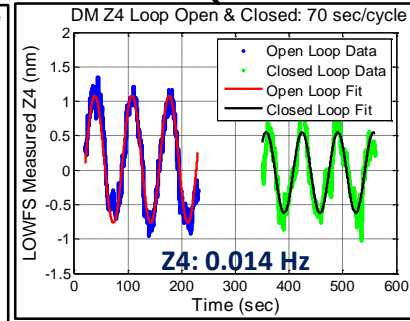
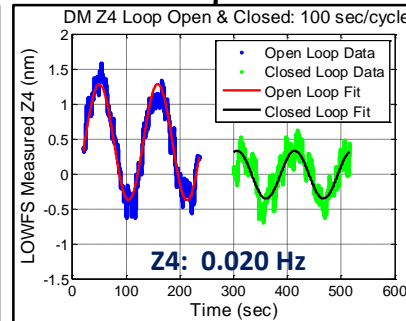
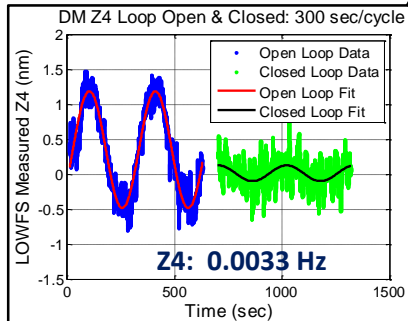
- Testbed source focus position is driven in sinusoidal fashion, creating sinusoidal focus swing of about  $\pm 2$  nm.
- LOWFS sensor measured focus and other low order modes.
- **LOWFS focus measurement (smoothed or fitted) remains fair constant through many orders of magnitude ( $10^4\times$ ) of source brightness**



- Focus drift generated by OTA simulator
  - $\pm 2$  nm swing sinusoidal focus disturbance
  - 4X larger than expected in WFIRST flight
- DM is used to correct focus (Z4)
- Solid blue line is the model prediction and green squares are testbed data
- Open (blue) and closed loop (green) LOWFS Z4 measurement at various frequencies are plotted in the sub-panels.



- Excellent agreement between model and TB measured DM loop performance



- LOWFS/C performance has been tested using source brightness equivalent to stars in the flight like condition ( $M_v \leq 5.0$ ).
- With WFIRST like line-of-sight jitter injected by the testbed's Jitter Mirror the LOWFS/C can maintain the contrast stability for source as faint as  $M_v = 5$ . The post correction residual jitter, measured by coronagraph contrast, has shown to meet the WFIRST jitter requirement of 0.5 mas.
- Using source equivalent to  $M_v = 5$  the testbed measured focus error rejection matches model prediction very well for LOWFS/C low order (focus) correction loop using a DM.
- We have also demonstrated simultaneous starlight light suppression wavefront control (EFC) while LOWFS/C is correcting the injected WFIRST like line-of-sight and wavefront disturbances, which will be reported in the next talk.
- Future work on the testbed for LOWFS/C:
  - Integrating an existing integral field spectrometer (IFS) to the OMC testbed to demonstrate CGI spectroscopy mode working with LOWFS/C under dynamic condition.
  - Updating OTA-Simulator which includes updated jitter mirror and pinhole relay optics which will provide more capability of dynamic wavefront test
  - New CGI mask designs

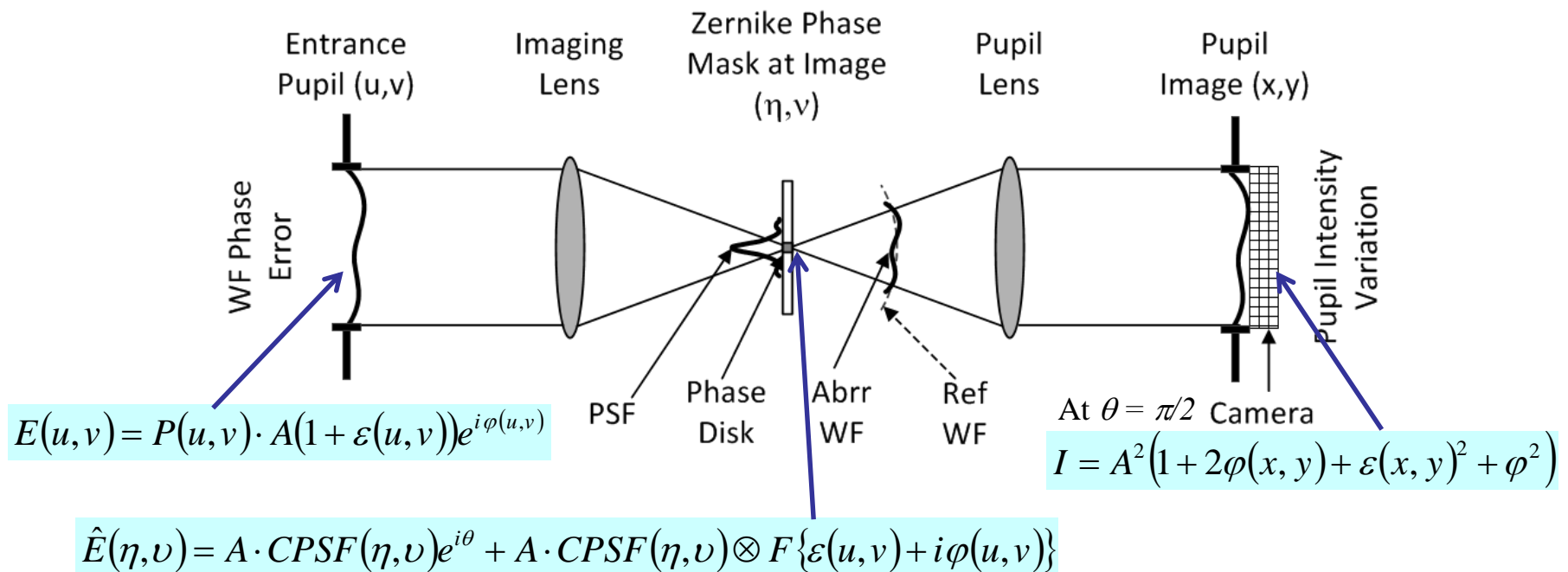


# Backup Slides



# Zernike Wavefront Sensor Concept

- **Zernike WFS (ZWFS) measures wavefront error (WFE) from interference between the PSF light passing through inside and outside the phase dimple (diameter  $\sim \lambda/D$ ) placing at the PSF core**
  - Same principle as Zernike phase contrast microscope
  - With phase dimple at phase shift of  $\pi/2$ , pupil image intensity variation is proportional to the WFE:  $\Delta I \sim \pm 2\phi$
- **WFIRST CGI LOWFS uses linearized differential image to sense the delta WFE**
  - Rejected starlight from  $\sim 3 \lambda/D$  focal plane mask cause the ZWFS can only measure low order WFE
  - LOWFS camera samples pupil at  $32 \times 32$  pixels with 20% band light to improve ZWFS' SNR
- **ZWFS converts pupil phase error into intensity variation on the LOWFS camera**



- **Line-of-sight control uses both feedback and feedforward loops**
- **Feedback path to cancel slow ACS LoS drift**
  - LOS loop is shaped for **optimal rejection of the ACS disturbance and LOWFS/C sensor noise**. This is done by balancing the error contribution from camera noise and LoS drift from ACS
- **Feedforward path to cancel high frequency tonal LoS jitter from reaction wheels**
  - RWA speed information is used to determine the disturbance frequencies
  - A least-mean-square (LMS) filter estimates the gain and phase of the tonal disturbances
  - Correction commands are directly sent to FSM

